

The Features Vector Research on Target Recognition of Airplane

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Abstract

The selection of the features vector is crucial, taking direct effect on the accuracy of target recognition.

Considering that the airplane has smooth surface and regular geometric shape, this paper chooses the geometric shape feature to describe the target of airplane. These geometric features vector include the center point, the characteristics of size and shape, the ration of airframe-length to wingspan etc.

Finally, this paper applies Support Vector Machine to validate the effectiveness of geometric features vector. From the result of experiment, it can be concluded that these features vector can satisfy the requirement of target recognition.

Keywords: Target Recognition; Features Vector; Geometric Features; Support Vector Machine (SVM)

1. Introduction

For the field of image processing, feature extraction and target recognition have a close relation. Feature extraction is based on the research between target and environment characteristic, aim to extract the features expressing the essence attribute of target, these features vector that make it possible to apart the target from environment. The target recognition is based on pattern recognition technology, aim to apart the target form background according to the target features vector

that are obtained by feature extraction process, furthermore to ascertain the type of target, its location and another useful information, etc.

The image feature is both the basic content of inner image and the most essential attribute hold by itself and distinguish itself from the other image. The image feature can be natural attribute that human vision can discern, called by nature feature, such as brightness, color, outline, structure etc. The image feature can be man-made attribute, called by man-made feature, normally, the man-made feature is parameter vector that is measured and computed by the researcher, for example, the histogram, the frequency spectrum, the area, and the perimeter etc. In this paper, mainly discuss the man-made feature of the airplane image.

2. Feature Extraction on Airplane Target

Normally, the feature of image can be classified for seven groups: (1) the scope feature; (2) statistics feature; (3) transformation feature; (4) boundary feature; (5) topological structure feature; (6) texture feature; (7) geometry feature.

Among of these features, the geometry feature takes an important status for target recognition. To pattern recognition and computer vision, the geometry is the intrinsic characteristic of object. Using the geometry of object can obtain other attributes, such as the borderline of object, the surface normal line etc. At the same

time, the geometry of object can be directly sense by humane visions, so it is easier to extract and process.

The airplane has smooth surface and its geometric shape has certain regulation. Contrast to the outstanding geometric information, texture and the other information are weakness. Thus this paper chooses the geometry feature to describe the target of airplane.

2.1. The Center Point

According to practically situation, the target of airplane is impossible to be a single dot in image. Thus, this paper takes the center point of the area to depict the location of the object. The center of the area is defined as O, the center of mass for the image with the same shape and the same quality in unit area. For the binary image $f(x,y)$, if the location of the target's pixel is (x_i, y_i) , here, $i=0,1, \dots, N-1, j=0,1, \dots, M-1$, then the center of the target can be computed by :

$$\begin{aligned}\bar{x} &= \frac{1}{N * M} \sum_{i=0}^{N-1} \sum_{j=0}^{M-1} x_i \\ \bar{y} &= \frac{1}{N * M} \sum_{i=0}^{N-1} \sum_{j=0}^{M-1} y_j\end{aligned}\quad (1)$$

2.2. The Characteristic of Size

2.2.1 The Perimeter and the Area

Normally, the perimeter of the object can be defined for the boundary length of the target. Thus, for the digital image, the perimeter L can be computed by the total number of pixels at the boundary of the target. After the border track for the target, it may be hypothesized that the digital curve outline is $P=\{p_1, p_2, p_3, p_4, \dots, p_n\}$, here, n is the number of the pixels on the curve outline, so the perimeter L is equal to n, namely:

$$L = n$$

Subsequently, the parameter of the area is convenient to obtain too. The area

of the target is related to the boundary, not to the interior gray level. The easiest method of computation is to count the number of pixels within or on the boundary of the target. For a image with dimension $N * M$, if the gray distribution of the target is depicted by $f(x,y)$, the area of the target can be computed by:

$$A_{area} = \sum_{i=1}^N \sum_{j=1}^M f(x_i, y_j) \quad (2)$$

To binary value image, supposed that the value of 1 is the object, and 0 for background, the computation of area is to count the number of those pixels that the gray level is 1.

2.2.2 Macro Axis and Minor Axis

When the boundary of the object is known, it is the most simple and effective method to depict the basic shape with the size of the enclosing rectangle. The enclosing rectangle on coordinate orientation can be represented by the span on the horizon and vertical. Aim to the span, just computing the maximum and minimum of the points on the boundary curve. But for the target at random orientation, the horizon and vertical is not crucial. Thus it is necessary to determine the principal axis of the target. By computing the minimum enclosing rectangle, the principal axis can be obtained.

The basic idea of computing the MER(minimum enclosing rectangle) is rotating the target by the increment of three angles or five angles, within 90 angles. Every rotating, record the maximum and minimum of x and y of the enclosing rectangle on this coordinate orientation. Rotating to α angles, the area of the enclosing rectangle is the smallest. Here is MER. And this orientation is the principal axis.

Under this situation, we can compute the macro axis and minor axis. Namely, the macro axis is the width of MER, the minor axis is the height of

MER. The computation can be depicted by:

$$\begin{aligned} \text{LongAxis} &= \text{MER}(\text{width}) \\ \text{ShortAxis} &= \text{MER}(\text{height}) \end{aligned} \quad (3)$$

2.3. The Characteristic of Shape

2.3.1 Compactness

Compactness reflects the proportion of the target to its enclosing rectangle. Namely, compactness is the value that the pixels of the target divide into the pixels

of the MER. As above, the A_{MER} (the area of the MER) is equal to the number of pixels of MER. So, compactness can be depicted by:

$$J = \frac{A_{area}}{A_{MER}} \quad (4)$$

Here, A_{area} is the area of the target, A_{MER} is the area of the MER.

2.3.2 Complexity

Complexity is defined as the ration of the number of pixels on the boundary curve to the number of pixels of the wholly target. For the digital image, the number of pixels on the target's boundary is equal to the perimeter L, and the number of the wholly target's pixels

is the area A_{area} , so, complexity of the target can be computed by:

$$F = \frac{L}{A_{area}} \quad (5)$$

2.4. The Ration of Airframe-length to Wingspan

Aim to obtain the ration of airframe to wingspan, it is necessary to ascertain the direction of the principal axis. The principal axis of airplane is defined as a line that makes the moment of inertia of the region D the minimum. The moment of inertia of region D can be computed by:

$$\begin{aligned} I &= \iint_D [(x - \bar{x}) \sin \alpha - (y - \bar{y}) \cos \alpha]^2 f(x, y) dx dy \\ &= \sin^2 \alpha \iint_D (x - \bar{x})^2 f(x, y) dx dy \\ &\quad + \cos^2 \alpha \iint_D (y - \bar{y})^2 f(x, y) dx dy \\ &\quad - 2 \sin \alpha \cos \alpha \iint_D (x - \bar{x})(y - \bar{y}) f(x, y) dx dy \end{aligned} \quad (6)$$

Here, α is inclination of the principal axis through the center (\bar{x}, \bar{y}) with the x axis.

As above, the center (\bar{x}, \bar{y}) can be computed by Eq.1. And, the integral computation in eq. 6 can be deduced by central moment formula:

$$\mu_{pq} = \int_{-\infty}^{+\infty} \int_{-\infty}^{+\infty} (x - \bar{x})^p (y - \bar{y})^q f(x, y) dx dy \quad (7)$$

Therefore, the moment of inertia of region D may be simplified by:

$$I = \mu_{20} \sin^2 \alpha + \mu_{02} \cos^2 \alpha - 2\mu_{11} \sin \alpha \cos \alpha \quad (8)$$

Want to make the value of I the minimum,

$$\frac{dI}{d\alpha} = 0$$

then $\frac{dI}{d\alpha} = 0$, namely,

$$\mu_{20} \sin 2\alpha - \mu_{02} \sin 2\alpha - \mu_{11} \cos 2\alpha = 0 \quad (9)$$

Thereby,

$$\tan 2\alpha = \frac{2\mu_{11}}{\mu_{20} - \mu_{02}} \quad (10)$$

Then :

$$\alpha = \frac{1}{2} \tan^{-1} \left(\frac{2\mu_{11}}{\mu_{20} - \mu_{02}} \right) \quad (11)$$

After having the inclination α of the principal axis with the x axis, the airframe-length is parallel with the x axis, and the wingspan is parallel with the y axis. The airframe-length x' is the projection of the airplane to the x axis, and the wingspan y' is the projection of the airplane to the y axis. Therewith, the ra-

tion of the airframe-length to wingspan is computed by $T = x' / y'$.

3. Experiment

3.1. The Result of Features Vector

Fig.1 is the top view of six types of airplane. According to equation of features vector above introduced, we can obtain some data of these images. Now, the experiment data are reported as table 1 and table 2.



(a) F/A-18 (b) F-22



(c) F-14 (d) F-15



(e) SU-27 (f) Mig-29

Fig. 1 the top vies of six types of airplane

From table 2, SU-27 and F-15 presented negative value. It is normal, because we prescribe that the right direction of the x axis is positive direction, from images, it is easy to find out that the course of two type of airplane is reverse to the right direction of the x axis. So the direction of the course is negative. From table 1 and 2, we find out that different type of airplane has different characteristic of shape, especially, the difference of the ration of airframe-length to wingspan

is very distinct. Obviously, the ration of airframe-length to wingspan is an important feature vector.

Table1. Experimental data

		F/A-18	F-22	F-14
center point	x	202.888	193.260	226.752
	y	139.221	149.471	157.429
Direction		0.023	0.207	0.474
Perimeter		1080	921	1248
Area		22243	19381	29347
Macro axis		393	401	380
Minor axis		158	140	226
Compactness		0.3582	0.3756	0.2681
Complexity		0.0486	0.0475	0.0437
Airframe-length/wing span		2.4873	2.8643	1.6978

Table2. Experimental data

		SU-27	F-15	Mig-29
center point	x	258.133	248.620	200.904
	y	148.006	174.770	167.698
Direction		-0.076	-0.437	0.087
Perimeter		955	1145	1121
Area		17699	21428	25885
Macro axis		361	350	436
Minor axis		134	227	209
Compactness		0.3725	0.2806	0.2972
Complexity		0.0540	0.0534	0.0444
Airframe-length/wing span		2.6940	1.5419	2.1373

3.2. Validating by SVM

There are three steps to validate the effectiveness of features vector by SVM: (1) modeling phase, (2) learning phase, (3) predicting phase.

Commonly, the model of SVM to recognize the target image is illustrated by Fig.2.

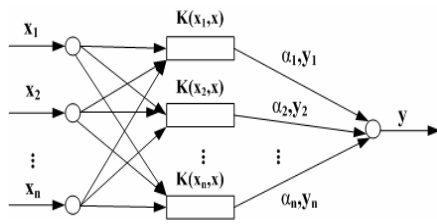


Fig.2 the model of SVM

In Fig.2, $K(x_i, x_j)$ is called by Kernel Function. At present, most SVM select RBF as Kernel Function. RBF is a function of exponent:

$$K(x \cdot x_i) = \exp\left(\frac{-\|x - x_i\|}{2\sigma^2}\right) \quad (12)$$

In this paper, we choose the software package LIBSVM to complete train and prediction, taking three-quarter data of total sample space to compose the training sample set, aim to train, taking a quarter data to compose the test sample set, aim to predict. The parameter configuration of LIBSVM and experiment result is reported by table 3.

Table3. Result by LIBSVM

Total sample	Kernel function	Training sample	Testing sample	Accuracy(%)
45	RBF	36	9	85.7143
80	RBF	60	20	86.0125
125	RBF	94	31	87.2617

4. Conclusion

See from Table 3, the accurateness is tend to increase by enhance the number to total sample. In case that the condition of experiment is allowable, we can further enhance the accurateness of prediction. But the rate of recognition 80% can illu-

minate that these features vector is effective.

5. References

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