

Mechanical structure optimization design of an

offshore backfilling plough -fenders

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Abstract

Offshore backfilling plough are used for backfilling for laying pipelines in seabed trenches; fenders are the primarily consist of an offshore backfilling plough; Research on the virtual join forces of the fenders is essential for predicting the join forces of the backfilling plough during backfilling. In this work, uniflow Fluid-structure interaction is developed to investigate the Join Forces of fenders under different mechanical structure. To analysis of this resulting data, a new fenders structure was carried out.

1 Introduction

Offshore pipelines on the seabed laid by a vessel are prone to damage by unexpected artificial and natural disturbances caused by fishing nets, anchors, wave oscillations, and other seabed features and seism. Hence, pipelines are normally

Beam Adjust frame Adjust frame

Fig.1. 3-D overview of the offshore backfilling plough

buried in trenches using offshore trenchers to prolong their service life and increase the pipelines stability in sea environment. Some of the work was done by backfilling plough.

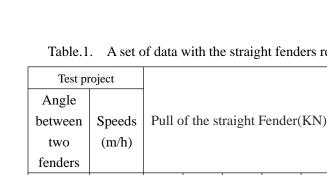
Mechanical backfilling equipment has been developed based on different pipeline installations and backfill methods in variety of sea and areas and conditions. Consequently, three type of mechanical backfill are classified according to their mode of backfilling as fender mode, fender-reamer mode and jetter mode. Fenders are used both in fender mode and fender-reamer mode.

The offshore backfilling plough consists of four components, as shown in Fig.1. Skid was used to support the system and guide the moving way and adjust the cut depth; Adjust frame was designed to adapt different grooves; Beam to support the overall system. Collect dirt and transport the dirt to grooves was the main work of fenders. When working, the stress of the fenders was the main source of the backfilling plough. Mechanical structure optimization design of the fenders was the most important work to save the system power demand.

An onshore field experimental verification was done, as shown in Fig.2. the model was done in 1:6, through agricultural soil experiment, one of the straight soil, one of



Fig.2. Onshore field experimental



5.12

6.34

200

360

PRESS

64°

Table.1.	A set of data	with the straight fend	ers results
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5.63

6.49

6.10

6.56

5.69

6.64

5.58

6.83

5.49

6.84

the straight fenders results are shown in Table.1. From the results show that the fenders need too much energy to move. The onshore field is that all, What about offshore? Can the mechanical structure of fenders be changed which would save any energy?

Different mechanical structure model of the 2 fenders

The system was designed to working offshore 500m, the mechanical structure model was simplified to good in the analysis. Fig.3. is the Simplified model of the fenders for

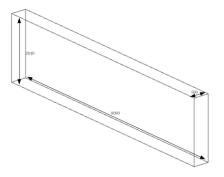
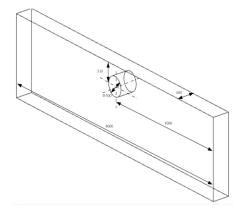
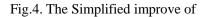
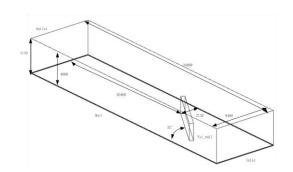
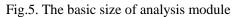


Fig.3. The Simplified model of fenders









analyze and Fig.4. is the Simplified improve model of the fenders. The analysis model of basic size according to the actual size. Fig.5. is the basic size of analysis model.

3 Selection of multiphase models

In this paper, the system was designed to working offshore. When the system work, the fenders push soil and water movement. So, there are two phases. In the simulation, for the sake of convenience, assume the fender is not moving; the water and mud rush to the fenders, and the mud would get to different fender heights; and all phases are incompressible; This is consistent with the Eulerian Model.

3.1 Volume fraction equation

The description of multiphase flow as interpenetrating continua incorporates the concept of phasic volume fractions. Volume fractions represent the space occupied by each phase, and the laws of conservation of mass and momentum are satisfied by each phase individually. The derivation of the conservation equations can be done by ensemble averaging the local instantaneous balance for each of the phases or by using the mixture theory approach.

Denoted α_q as volume fractions of the volume of phase, the volume of this phase is

$$V_q = \int_V \alpha_q dV \tag{1}$$

Where

$$\sum_{q=1}^{n} \alpha_q = 1 \tag{2}$$

The effective density of phase q is

$$\widehat{\rho}_q = \alpha_q \rho_q \tag{3}$$

where ρ_q is the physical density of phase q.

The volume fraction equation may be solved through explicit time discretization. The explicit formulation is time-dependent and the volume fraction is discretized in the following manner:

$$\frac{-\alpha_q^{n+1}-\alpha_q^n}{\Delta t}\mathbf{V} + \sum_f \left(U_f^{n+1}\alpha_{q,f}^{n+1}\right) = 0 \qquad (4)$$

where

n + 1 = infex for new (current) time step

n = infex for previous time step

 $\alpha_{q,f}$ = face value of the qth volume fraction

V = volume of cell

 U_f = volume flux through the face, based on normal velocity

3.2 Conservation of Mass

3.2.1 Continuity equation

The continuity equation for phase q is

$$\frac{\partial}{\partial t}(\alpha_q) + \nabla \cdot (\alpha_q \rho_q \overline{V_q}) = \sum_{p=1}^n (\dot{m}_{pq} - \dot{m}_{qp}) + S_q \quad (5)$$

where $\overrightarrow{V_q}$ is the velocity of phase q and \dot{m}_{pq} characterizes the mass transfer from the p^{th} to q^{th} phase, and \dot{m}_{pq} characterizes the mass transfer from phase q to phase p, and by the continuity equation, we could found that

$$\dot{m}_{pq} = -\dot{m}_{qp} \tag{6}$$

$$\dot{m}_{pp} = 0 \tag{7}$$

3.2.2 Conservation of Momentum

The momentum balance for phase q yields

$$\frac{\partial}{\partial t} (\alpha_q \rho_q \vec{V}_q) + \nabla \cdot (\alpha_q \rho_q \vec{V}_q \vec{V}_q) = -\alpha_q \nabla p + \nabla \cdot \bar{\bar{\tau}}_q + \alpha_q \rho_q \vec{g} + \sum_{p=1}^n (\vec{R}_{pq} + \dot{m}_{pq} \vec{V}_{pq} - \dot{m}_{qp} \vec{V}_{qp}) + (\vec{F}_q + \vec{F}_{lift,q} + \vec{F}_{wl,q} + \vec{F}_{vm,q} + \vec{F}_{td,q})$$
(8)

Where $\overline{\overline{\tau}}_q$ is the q^{th} phase stress-strain tensor.

$$\bar{\bar{\tau}}_q = \alpha_q \mu_q \left(\nabla \vec{V}_q + \nabla \vec{V}_q^T \right) + \alpha_q \left(\lambda_q - \frac{2}{3} \mu_q \right) \tag{9}$$

Here μ_q and λ_q are the shear and bulk viscosity of q, \vec{F}_q is an external body force, $\vec{F}_{lift,q}$ is lift force (described in lift force), $\vec{F}_{wl,q}$ is a wall lubrication force (described in wall lubrication force), $\vec{F}_{Vm,q}$ is a virtual mass force, and $\vec{F}_{td,q}$ is a turbulent dispersion force (in the case of turbulent flows only). \vec{R}_{pq} is an interaction force between phases, and p is the pressure shared by all phases. \vec{V}_{pq} is the interphase velocity.

Equation (7) must be closed with appropriate expressions for interphase force \vec{R}_{pq} . This force depends on the friction, pressure, cohesion, and other effects, and is subject to the conditions that $\vec{R}_{pq} = -\vec{R}_{qp}$ and $\vec{R}_{qq} = 0$.

A formulas Interaction term are used to represent as the following form:

$$\sum_{p=1}^{n} \vec{R}_{pq} = \sum_{p=1}^{n} K_{pq} \left(\vec{V}_{p} - \vec{V}_{q} \right)$$
(10)

Where $K_{pq}(=-K_{qp})$ is the interphase momentum exchange coefficient, and \vec{V}_p and \vec{V}_q are the phase velocities.

3.3 Interphase Exchange Coefficients

For fluid-fluid flows, each secondary phase is assumed to form droplets or bubbles. This has an impact on how each of the fluids is assigned to a particular phase. For example, in flows where there are unequal amounts of two fluids, the predominant fluid should be modeled as the primary fluid, since the sparser fluid is more likely to form droplets or bubbles. The exchange coefficient for these types of bubbly, liquid-liquid mixtures can be written in the following general form:

$$K_{pq} = \frac{\rho_{Pf}}{6\tau_P} d_{pA_i} \tag{11}$$

where A_i is the interfacial area.

where τ_p , the "particulate relaxation time", is defined as

$$\tau_p = \frac{\rho_{pd_p^2}}{18\mu_q} \tag{12}$$

where d_p is the diameter of the bubbles or droplets of phase p.

Where, the drag function, is defined by Schiller and Naumann Model, which is used to represent as

$$f = \frac{c_D R_e}{24} \tag{13}$$

Where

$$C_D = \begin{cases} 24 \left(1 + 0.15 R_e^{0.678} \right) / R_e R_e \le 1000 \\ 0.44 \quad R_e > 1000 \end{cases}$$
(14)

and R_e is the relative Reynolds number. The relative Reynolds number for the primary phase *q* and secondary phase *p* is obtained from

$$R_e = \frac{\rho_q |\overline{\nu}_p - \overline{\nu}_q| d_p}{\mu_q} \tag{15}$$

The relative Reynolds number for secondary phase p and r is obtained from



$$R_e = \frac{\rho_{rp} |\bar{\nu}_r - \bar{\nu}_p| d_{rp}}{\mu_{rp}} \tag{16}$$

where $\mu_{rp} = \alpha_p \mu_p + \alpha_r \mu_r$ is the mixture viscosity of the phases *p* and *r*.

3.4 Transport Equations for the Standard $k - \varepsilon$ Model

The turbulence kinetic energy, k ,and its rate of dissipation, ϵ ,are obtained from the following transport equations:

$$\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_i}(\rho k u_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + G_k + G_b - \rho \varepsilon - Y_M + S_k$$
(17)

and

$$\frac{\partial}{\partial t}(\rho\varepsilon) + \frac{\partial}{\partial x_i}(\rho\varepsilon u_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial\varepsilon}{\partial x_j} \right] + C_{1\varepsilon} \frac{\varepsilon}{k} (G_k + C_{3\varepsilon}G_b) - C_{2\varepsilon} \rho \frac{\varepsilon^2}{k} + S_{\varepsilon}$$
(18)

In these equations, G_k represents the generation of turbulence kinetic energy due to the mean velocity gradients, calculated as described in following:

$$G_k = -\rho \overline{u'_i u'_j} \frac{\partial u_j}{\partial x_i} \tag{19}$$

where, G_b is the generation of turbulence kinetic energy due to buoyancy, obtained from:

$$G_b = \beta g_i \frac{\mu_t}{Pr_t} \frac{\partial T}{\partial x_i}$$
(20)

where Pr_t is the turbulent Prandtl number for energy and g_i is the component of the gravitational vector in the i^{th} direction. For the standard and realizable $k - \varepsilon$ models, the default of Pr_t is 0.85.

 Y_M represents the contribution of the fluctuating dilatation in compressible turbulence to the overall dissipation rate, obtained from:

$$Y_M = 2\rho \varepsilon M_t^2 \tag{21}$$

where M_t is the turbulent Mach number, defined as:

$$M_t = \sqrt{\frac{k}{a^2}} \tag{22}$$

Where $a(\equiv \sqrt{\gamma RT})$ is the speed of sound.

 $C_{1\varepsilon}$, $C_{2\varepsilon}$, $C_{3\varepsilon}$, are constants. σ_k and σ_{ε} are the turbulent Prandtl number for k and ε . S_k and S_{ε} were defaulted.

4 Boundary Conditions

4.1 Velocity inlet

The mass flow rate entering a fluid cell adjacent to a

velocity inlet boundary is computed as

$$\dot{m} = \int \rho \vec{V} \cdot d\vec{A} \tag{23}$$

4.2 Pressure outlet

The flow exiting the boundary is subsonic, the strong Averaging approach was in application, the face pressure value P_f for subsonic exit flow is computed using the following expression:

$$P_f = P_e + (1 - F) (P_C - P_{C_avg})$$
(24)

where

 P_C = interior cell pressure at neighboring exit face, f $P_{c_{avg}}$ = averaged interior cell pressure

at a boundary

 P_e = specified exit pressure

F = pressure blending factor;

F = 0 recovers the fully averaged pressure,

F = 1 recovers the specified pressure.

4.3 Wall Motion

The velocity of the fluid at the wall boundary is set to zero (no moving, no wall velocity), so the boundary condition for the velocity becomes:

$$U_{Wall} = 0 \tag{25}$$

5 Equation solution

5.1 Conservation of Energy

The volume fraction of each phase is calculated from a continuity equation:

$$\frac{1}{\rho_{rp}} \left(\frac{\partial}{\partial t} \left(\alpha_q \rho_q \right) + \nabla \cdot \left(\alpha_q \rho_q \vec{V}_q \right) = \sum_{p=1}^n \left(\dot{m}_{pq} - \dot{m}_{qp} \right) \right)$$
(26)

where ρ_{rp} is the phase reference density, or the volume averaged density of the q^{th} phase in the solution domain.

5.2 Fluid-fluid Momentum Equations

The conservation of momentum for a fluid phase q is:

$$\frac{\partial}{\partial t} (\alpha_q \rho_q \vec{V}_q) + \nabla \cdot (\alpha_q \rho_q \vec{V}_q \vec{V}_q) = -\alpha_q \nabla p + \nabla \cdot \bar{\bar{\tau}}_q + \\ \alpha_q \rho_q \vec{g} + \sum_{p=1}^n (K_{pq} (\vec{V}_p - \vec{V}_q) + \dot{m}_{pq} \vec{V}_{pq} - \dot{m}_{pq} \vec{V}_{qp}) + \\ (\vec{F}_q + \vec{F}_{lift,q} + \vec{F}_{wl,q} + \vec{F}_{vm,q} + \vec{F}_{td,q})$$

$$(27)$$

Here \vec{g} is the acceleration due to gravity.

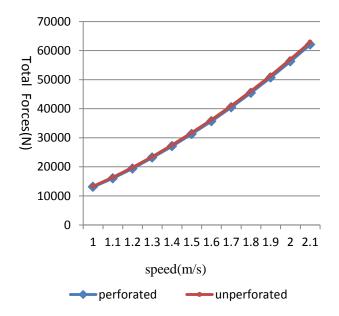


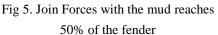
6 Case analysis

Primary Phase were water, Secondary Phase were mud. The mud flowed from the bottom and the water flowed from all. Here, for the mud,

Density, $\rho_1 = 1030(kg/m^3)$, Viscosity, $\mu_1 = 2.678(kg/m \cdot s)$; to the water, Density, $\rho_2 = 998.2(kg/m^3)$, Viscosity, $\mu_1 = 0.001003(kg/m \cdot s)$, Specific Heat $C_p = 4182(j/kg \cdot k)$, Thermal Conductivity $\lambda = 0.6(w/m \cdot k^2)$

Thermal Conductivity $\lambda = 0.6(w/m \cdot k)$ Molecular WeMr = 18.0152(kg/kgmol), Standard State EnthalpyS = -2.858412e8(j/kgmol), Standard State Entropy $Sm = 69902.21(j/kgmol \cdot k)$, L-J Characteristic Lengths_i = 1(angstrom), L-J Energy Parameter e/k = 100(k)Latent Heat $h_{fg} = 2263073(j/kg)$ Vaporization Temperature $T_{vap} = 284(k)$, Boiling Point $T_{bp} = 373(k)$, Volatile Component Fraction $f_{vD} = 100(\%)$, Droplet Surface Tension $\sigma = 0.0719404(n/m)$.

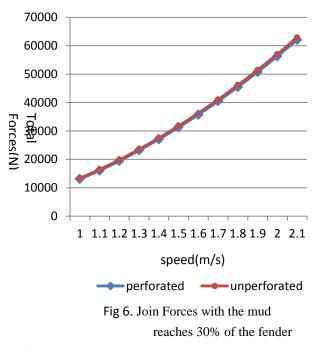




and the resulting data. The inlet velocity was from $1\sim 2.1(m/s)$, increasing by 0.1. The outlet pressure was $5MP_a$. The Tables all show that, the unperforated fender is

more stressed than the perforated fender, the mud reached the fender 30% was more obvious than 50%. With the increase of the inlet velocity, the Join Forces of the fender were approximately similar Straight line up. And for the mud to got to the fender 50%, when the velocity was changing from $1.1 \sim 1.4(m/s)$, the difference was more obvious; in particular, when the speed was 1.1 m/s, it reduces the force of 5.3%. Similarly, for the mud reached 30% of the fender, the overall reduction was close to 1.3%. When the inlet velocity was at the same time, the Join Forces of the fender has changed little with different mud position. These data showed that under the working conditions, the water is relatively large.

Compare table.1. with Fig.5. and Fig.6., the data showed that when the fender was working underwater, the Join Forces of the fender was less than which on land. The main reason is that when working underwater, the fender would be pushed by the water which reduced the Join Forces of the fender.



7 Conclusion

This paper presents a virtual 3-D uniflow fluid-fluid coupling simulation model of an offshore fender in WORKBENCH. The analysis of the different inlet velocity with different mud position. The analysis of the fender provides the Eulerian Model. Based on this model, several simulation analysis were carried out to predict the fender mechanical structure. The conclusion in this paper are follows:

(1) The analysis was consistent with the hypothesis. A perforated fender is less stressed than a unperforated which shows that it is helpful to reduce the Join Forces by increasing the number of holes in the case of the structural strength and the height of the mud.

(2) When the inlet velocity was 2.1(m/s), the Join Forces of the unperforated fender was 62860N where the perforated fender was 62081N. And when the inlet velocity was 1(m/s), the Join Forces of the unperforated fender was 13332N where the perforated fender was 13088N. A perforated structure is more efficient than a non-porous structure.

(3) When the mud reaches 50% of the fender, the inlet velocity was 1.1(m/s), the Join Forces reduction was more pronounced; consider the mud reaches 30% of the fender, relatively, a non-porous structure could save more energy with this speed.

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